

# Fundamental Issues of Lean Premixed H<sub>2</sub>/air Combustion for Gas Turbine Development

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# Objective & Approach of LBNL DOE-FE Project

- **Objective**

- ▶ Explore the feasibility of the low-swirl combustion concept for lean premixed syngas and H<sub>2</sub> turbines

- **Approach**

- ▶ Adapt low-swirl combustion for energetic H<sub>2</sub> fuels
  - Scale up nozzle design to larger sizes & throughputs
  - Optimize swirler for syngas and H<sub>2</sub> operation
  - Consider fuel-flexible options
- ▶ Address flashback and auto-ignition risks
  - Develop fuel injection and mixing strategies
  - Investigate turbulent flame speed and flashback mechanism

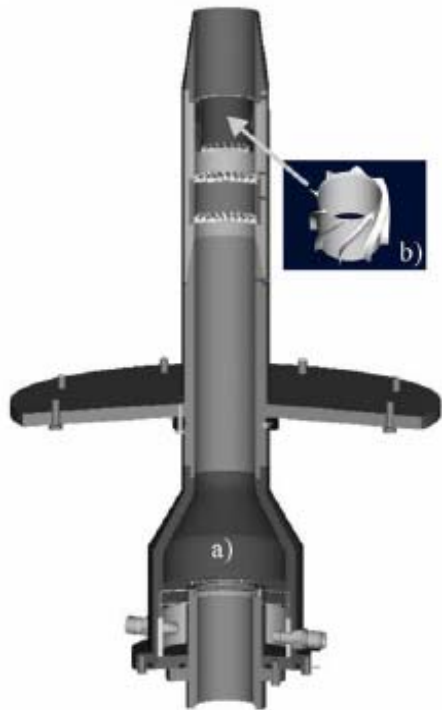
# Current Status

- **Developing a first-order analytical model for the turbulent flame brush position**
  - ▶ Linking swirl number to flame position, flowfield features, turbulent flame speed, and turbulence
  - ▶ Foundation for assessing performance and guiding hardware design
- **Investigated H<sub>2</sub>/air flames at STP**
  - ▶ Optimized LSI to operate with pure and diluted H<sub>2</sub> reactants
    - Requires a slightly lower swirl number than for hydrocarbons
  - ▶ Found linear turbulent flame speed correlation for H<sub>2</sub>
- **Testing with high H<sub>2</sub> fuels at NETL**
  - ▶ Applied knowledge from laboratory studies in the prototype design
- **Preparing tests with syngas and H<sub>2</sub> at Georgia Tech**
- **Initiated collaboration with Siemens Power Generation**

# Scientific Foundation Needed for Hardware Development

- **Basic understanding of the overall flame and flowfield behaviors at turbine conditions**
  - ▶ **Validate first-order analytical model and establish engineering guidelines for  $H_2$  and syngases**
    - **Limited knowledge of LSC compared to the wealth of information on high-swirl combustion**
  - ▶ **Gain physical insights to address system integration issues and reduce design iterations**
- **Detail characteristics of laminar and turbulent flames**
  - ▶ **Develop turbulent flame model for lean premixed syngas and  $H_2$** 
    - **Characterizing fuel effects on local heat release**
  - ▶ **Support development of predictive computational tools**
    - **High fidelity numerical simulations with detail chemistry that capture the entire flowfield, the flame, and its emissions**
      - **Limited capability of RANS CFD due to its averaged nature**
      - **Standard LES based on typical flame models not applicable**

# Low-Swirl Burner Is An Effective Tool to Address Fundamental Issues



**LBL/Lund/Darmstadt LSB  
for fundamental premixed  
turbulent flame studies**

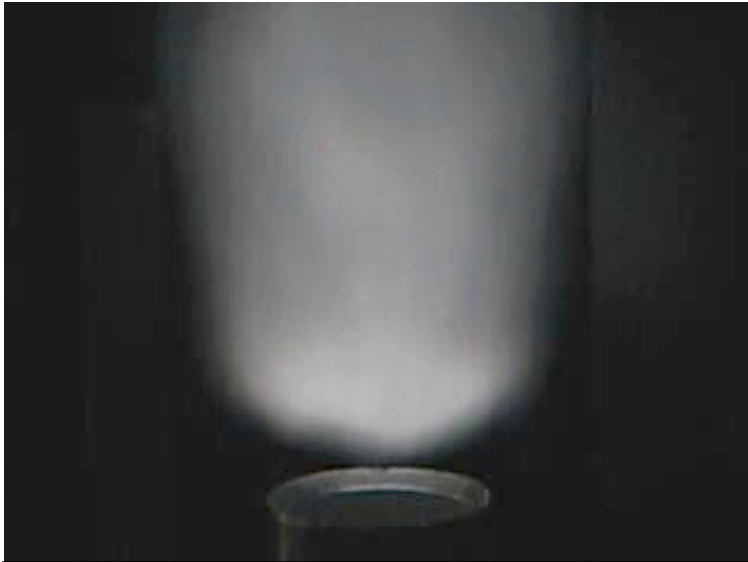
- Provides a close approximation of a locally planar 1D premixed turbulent flame
  - ▶ Adopted by researchers world-wide
- Laboratory data directly relevant to gas turbine development
  - ▶ Research burners and gas turbine hardware have the same configuration
  - ▶ Laboratory flame properties and behaviors similar to those observed at turbine conditions
    - Better understanding of this link will greatly facilitate development for H<sub>2</sub> turbines

# Fundamental Issues For LSC Development

- Ignition delay
- Combustion dynamics
- LSC flowfield evolution with velocity, temperature & pressure
- Turbulent flame speed
- Properties of lean laminar syngas and H<sub>2</sub> flames
- Chemical kinetics (molecular transport and rate coefficients) of lean H<sub>2</sub> and syngas systems
- Heat release models for H<sub>2</sub> and syngas turbulent premixed flames at relevant turbine conditions
- NO<sub>x</sub> formation
- Turbulence models for swirling flows
- Computational methods for turbulent reacting flows

# Issues Being Addressed by Current Project

# Laboratory Experiments on Premixed H<sub>2</sub> Flames



Lean premixed H<sub>2</sub>/air flame at  
 $\phi = 0.35$  &  $U_0 = 20$  m/s  
captured by an IR camera

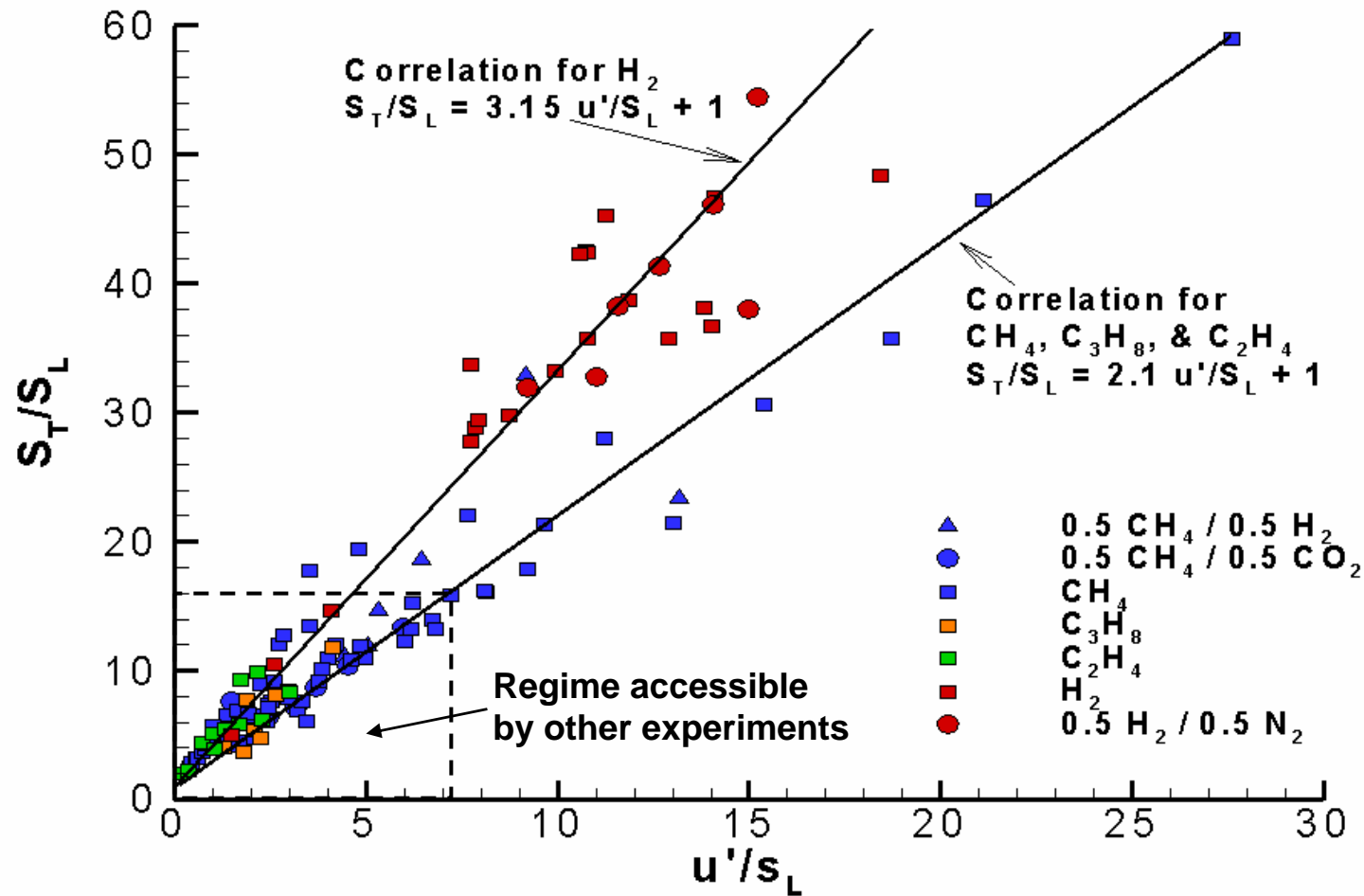
- Provide basic knowledge to configure LSC for H<sub>2</sub> turbine
- Studied flames burning with pure H<sub>2</sub> and H<sub>2</sub> diluted with 25% and 50% N<sub>2</sub> dilution (by volume)
  - ▶ Conditions with  $T_{ad} = 13800, 1320, 1200$  K
- Varied swirl number from  $0.57 < S < 0.41$ 
  - ▶ Varied the blockage ratio and geometric pattern of the center plate
- Measured lean blow-out and flowfield characteristics at  $9 < U_0 < 20$  m/s
- Found optimum configuration for H<sub>2</sub>
- PIV measures  $S_T$  and other flowfield characteristics



# Turbulent Flame Speed Central to the Low-swirl Combustion Concept

- Lifted flame exemplifies propagating nature of premixed combustion
- Correlation of turbulent flame speed,  $S_T$ , with turbulence intensity provides empirical constant for the LSC analytical model
- 4 different ways to define turbulent flame and each has its own specific meaning:
  1. **Local displacement flame speed – most relevant to LSC**
  2. Averaged displacement flame speed – bomb experiments
  3. Local consumption flame speed
  4. Averaged consumption flame speed – conical flames
- ▶ Comparison and interpretation of  $S_T$  from different definitions are counter-productive
- Local turbulent displacement flame speed at the axis most important to LSC
  - ▶ Flame speed at the stabilization point and flashback origin
    - Laboratory measurements and correlation of  $S_T$  give insights into flashback, fuel effects, and assist the design of fuel-flexible swirlers and injectors
    - Lacking data at turbine conditions

# $S_T$ of $H_2$ Flames Show Linear Correlation



# Implication of $S_T$ Correlation

- **Local turbulent displacement flame speeds measured in low-swirl combustion systems are significant higher than the averaged displacement flame speeds measured in non-stationary flames and the averaged consumption flame speed measured in stationary flames**
  - ▶ Averaged flame speed inadequate for assessing flashback risks
  - ▶ Local flame speed at the flashback origin is most useful
- **50% difference in  $H_2$  and  $CH_4$  turbulent displacement flame despite significant difference in laminar flame speed**
  - ▶ Correlation equations show contribution from laminar flame speed becoming insignificant at high  $u'$

# Need H<sub>2</sub> Turbulent Flame Speed Correlation at Relevant Gas Turbine Conditions

- **Significant experimental challenges for laser anemometry measurements in high-pressure flow channels**
  - ▶ Transmitting laser and scattered signal through windows reduces signal to noise
  - ▶ Operational difficulties associated with cleanup of fine particles deposited on windows
- **University and National Lab researchers rising to this challenge**
  - ▶ Overall trend can be inferred by observing LSI H<sub>2</sub> flames at high T and P
  - ▶ Detailed measurements at selected data points for quantitative comparison with results from STP laboratory experiments

# Open Issues

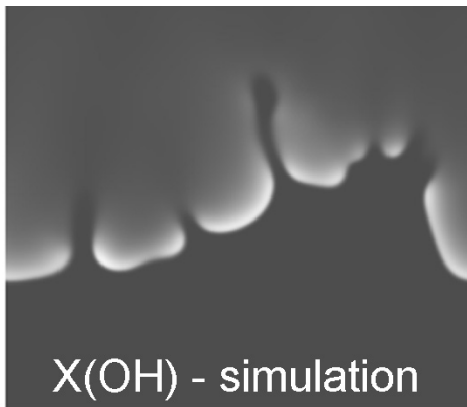
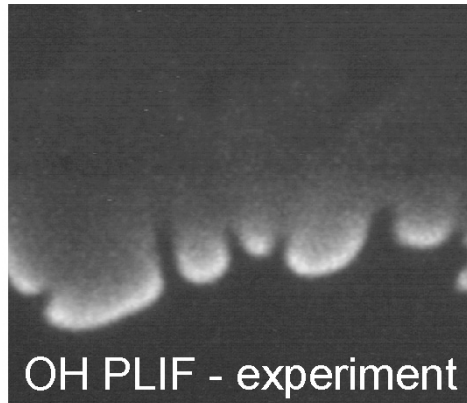
# Limited Data Available For Lean H<sub>2</sub> Laminar Flame Speeds

- Laminar H<sub>2</sub>/air premixed flames are inherently unstable for  $\phi < 0.7$  due to thermo-diffusive Instability
  - ▶ Cellular flame structures studied experimentally and theoretically
- Instability affects measurements of laminar flame speed  $S_L$ 
  - ▶ Bomb experiments requires corrections to obtain un-stretched laminar flame speed
  - ▶ Counterflow burner generates positive stress to suppress thermo-diffusive instability
- Chemical kinetics schemes for lean H<sub>2</sub>/air flames under predict measured  $S_L$ 
  - ▶ Huge scatter in predicted  $S_L$  of lean H<sub>2</sub> flames
  - ▶ Rate constants for third body reactions and molecular transport coefficients need further studies
- Large uncertainties exist for a basic property needed to analyze LSC flame processes
  - ▶ Affects the accuracy of the  $S_T$  correlation coefficient but not its linearity

# Reaction Front Instability Affecting $\text{H}_2$ /Air Turbulent Premixed Flame

- Turbulence produces both positive and negative stretch
  - ▶ Suppression and promotion of reaction front instability control how  $\text{H}_2$  flames respond to turbulence
  - ▶ Variations in reaction front structures and its wrinkle topology dictate local heat release rates and its spatial distribution
- Pressure increases thermo-diffusive instability
  - ▶ Coupling of pressure & turbulence effects remains unexplored
- Turbulence time scales can be much faster than preferential transport time scales
  - ▶ Can high turbulence suppress thermo-diffusive instabilities?
- Significant implication on turbulent flame model
  - ▶ Stretched laminar flamelet model cannot be applied to lean  $\text{H}_2$  turbulent flames

# Understanding and Characterizing Lean H<sub>2</sub>/Air Premixed Turbulent Flames



- **Research Need**

- ▶ Evolution of preferential diffusion effects with turbulence intensity, temperature and pressure
- ▶ Development of heat release model for LES and CFD

- **Opportunity**

- ▶ Coordinated experimental and computational investigations of premixed turbulent flame structures
  - Preliminary results obtained from OH-PLIF measurement and 3D time dependent simulation of H<sub>2</sub>/air flames in LSB
    - Wrinkle topology is drastically different than those of hydrocarbon flames
    - Non-uniform heat release with significant enhancement and local extinction due to thermo-diffusive instability are evident on the experimental and computational results



# Characterizing H<sub>2</sub>/Air Reaction Zone Structures

- **Research Needs**

- ▶ Understand the different responses to positive and negative stretch
  - Only positively stretched laminar flames have been studied
  - Very limited knowledge on negatively stretched flame with local extinction
- ▶ Improve species transport in chemical kinetic scheme

- **Opportunity**

- ▶ Controlled 2D oscillating laminar flame experiment generating regular and stationary flame wrinkles for detailed interrogation by laser methods
  - Stationary flame wrinkle structures amenable to diagnostics
  - Ideal for low Mach Number simulation with adaptive mesh refinement
- ▶ Test bed for studying effects of preferential diffusion on unsteady flame characteristics and chemistry/turbulence interactions
  - 2D flowfield reduces experimental ambiguities for direct comparison with simulations

# Predicting NO<sub>x</sub> Emissions at Turbine Conditions

- **Research Need**

- ▶ NO<sub>x</sub> emissions difficult to predict
  - Temperature and pressure dependencies of N<sub>2</sub> chemistry
  - Log linear dependence of LSC NO<sub>x</sub> on T<sub>ad</sub> regardless of fuel, T and P
    - Why does this dependency exists?
    - When does this dependency breakdown?

- **Opportunity**

- ▶ Need improved understanding of NO<sub>x</sub> formation in ultra-lean flames and the influence of the flowfield
- ▶ Experimental and computational study to obtain a scientific underpinning
  - Comparing NO<sub>x</sub> contours in high-swirl and low-swirl combustors
  - Correlate recirculation zone strength with NO<sub>x</sub> concentrations
  - Compute NO<sub>x</sub> dependency on residence time at STP and at high T and P

# Combustion Dynamics

- **Research Needs**

- ▶ **Response of LSC to external excitations**

- **Lifted flame and absence of strong recirculation affect the responses**
      - **LSI in T70 engine shows different acoustic signature**
      - **Acceptable pressure fluctuations at frequencies different than SoLoNOx high-swirl injectors**

- **Opportunity**

- ▶ **Side-by-side comparison of low-swirl combustion and high-swirl combustion subjected to the same acoustic forcing and fuel/air fluctuations**

- **Fresh insights to understand combustion dynamics**
    - **Model for LSC oscillations**
    - **Collaboration with Universities and National Laboratories researchers**

# Merging Measurements, Chemistry, Models & Simulation

- **Research Need**

- ▶ Computational tools to predict LSC flame and flowfield properties and their coupling with combustor geometries

- **Opportunity**

- ▶ Establish a benchmark experimental and computational configuration
  - Side-by-side comparison of low-swirl and high-swirl combustion in a gas turbine simulator with well-defined boundary conditions
- ▶ Promote cooperation among experimentalists, computational scientists and tool developers
  - Evaluate capabilities and weaknesses of RANS, LES, LMC and DNS
  - Define experimental configuration with well defined input, side and exit boundaries
  - Obtain benchmark data to calibrate and improve the computational methods
  - Develop robust flame and turbulence models
- ▶ Improve predictive capability for LSC including flame volume and flame chamber interaction

# Summary

- Principle of LSC is described by a top level model that can be used to guide its scale-up for large utility turbines
  - ▶ Laboratory experiments are being performed to provide empirical inputs for hydrocarbons, syngas and hydrogen
- Development for syngas and H<sub>2</sub> turbines requires basic knowledge on the combustion properties of lean syngas and H<sub>2</sub> flames
  - ▶ Well-designed experiments and coordinated theoretical and numerical studies are critical to provide new insights on both **laminar and turbulent** premixed flames
    - Reaction zone structures and wrinkle topology of turbulent flames
    - Positively and negatively stretched laminar flames
  - ▶ Coordinate and merge knowledge from University and Nat'l Lab researchers
    - Ignition delay and combustion dynamics studies
    - Benchmark experiments at gas turbine conditions
    - Computational model for LSC
  - ▶ Share results with OEMs and vendors